

BEAM ELEMENT STIFFNESS MATRICES
CE 131 — Matrix Structural Analysis
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Local Element Flexibility Matrix: \mathbf{F} :

The element flexibility matrix, \mathbf{F} relates $\{\delta, \theta_1, \theta_2\}$ to $\{N, M_1, M_2\}$:

$$\begin{Bmatrix} \delta \\ \theta_1 \\ \theta_2 \end{Bmatrix} = \mathbf{F} \begin{Bmatrix} N \\ M_1 \\ M_2 \end{Bmatrix}$$

and can be obtained from Castigliano's Theorem.

First Column

Second Column

Third Column

$$N(x) = N \quad V(x) = \frac{M_1 + M_2}{L} \quad M(x) = M_1 \left(\frac{x}{L} - 1 \right) + M_2 \frac{x}{L}$$

The total potential energy of a beam with these forces is:

$$U = \frac{1}{2} \int_0^L \frac{N^2}{EA} dx + \frac{1}{2} \int_0^L \frac{M^2}{EI} dx + \frac{1}{2} \int_0^L \frac{V^2}{G(A/\alpha)} dx$$

By Castigliano's Theorem,

$$\delta = \frac{\partial U}{\partial N} = \int_0^L \frac{N(x) \frac{\partial N(x)}{\partial N}}{EA} dx = \int_0^L \frac{N(x) \cdot 1}{EA} dx = N \int_0^L \frac{dx}{EA}.$$

$$\begin{aligned} \theta_1 &= \frac{\partial U}{\partial M_1} \\ &= \int_0^L \frac{M(x) \frac{\partial M(x)}{\partial M_1}}{EI} dx + \int_0^L \frac{V(x) \frac{\partial V(x)}{\partial M_1}}{G(A/\alpha)} dx \\ &= \left(\int_0^L \frac{\left(\frac{x}{L} - 1 \right)^2 dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) M_1 + \left(\int_0^L \frac{\frac{x}{L} \left(\frac{x}{L} - 1 \right) dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) M_2 \end{aligned}$$

and

$$\begin{aligned} \theta_2 &= \frac{\partial U}{\partial M_2} \\ &= \int_0^L \frac{M(x) \frac{\partial M(x)}{\partial M_2}}{EI} dx + \int_0^L \frac{V(x) \frac{\partial V(x)}{\partial M_2}}{G(A/\alpha)} dx \\ &= \left(\int_0^L \frac{\frac{x}{L} \left(\frac{x}{L} - 1 \right) dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) M_1 + \left(\int_0^L \frac{\left(\frac{x}{L} \right)^2 dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) M_2 \end{aligned}$$

or, in matrix form,

$$\begin{Bmatrix} \delta \\ \theta_1 \\ \theta_2 \end{Bmatrix} = \begin{bmatrix} \int_0^L \frac{dx}{EA} & 0 & 0 \\ 0 & \left(\int_0^L \frac{\left(\frac{x}{L} - 1 \right)^2 dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) & \left(\int_0^L \frac{\frac{x}{L} \left(\frac{x}{L} - 1 \right) dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) \\ 0 & \left(\int_0^L \frac{\frac{x}{L} \left(\frac{x}{L} - 1 \right) dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) & \left(\int_0^L \frac{\left(\frac{x}{L} \right)^2 dx}{EI} + \int_0^L \frac{\alpha dx}{GAL^2} \right) \end{bmatrix} \begin{Bmatrix} N \\ M_1 \\ M_2 \end{Bmatrix}$$

For prismatic members, E , A , and I are constant along the length, and the element flexibility matrix is

$$\mathbf{F} = \begin{bmatrix} \frac{L}{EA} & 0 & 0 \\ 0 & \frac{L}{3EI} + \frac{1}{G(A/\alpha)L} & -\frac{L}{6EI} + \frac{1}{G(A/\alpha)L} \\ 0 & -\frac{L}{6EI} + \frac{1}{G(A/\alpha)L} & \frac{L}{3EI} + \frac{1}{G(A/\alpha)L} \end{bmatrix}$$

To neglect shear deformation, set $\alpha = 0$.

Local Element Stiffness Matrix: \mathbf{k}

The element stiffness matrix is the inverse of the element flexibility matrix, and for prismatic members,

$$\mathbf{k} = \begin{bmatrix} \frac{EA}{L} & 0 & 0 \\ 0 & \frac{(4+\Phi)EI}{(1+\Phi)L} & \frac{(2-\Phi)EI}{(1+\Phi)L} \\ 0 & \frac{(2-\Phi)EI}{(1+\Phi)L} & \frac{(4+\Phi)EI}{(1+\Phi)L} \end{bmatrix}$$

where

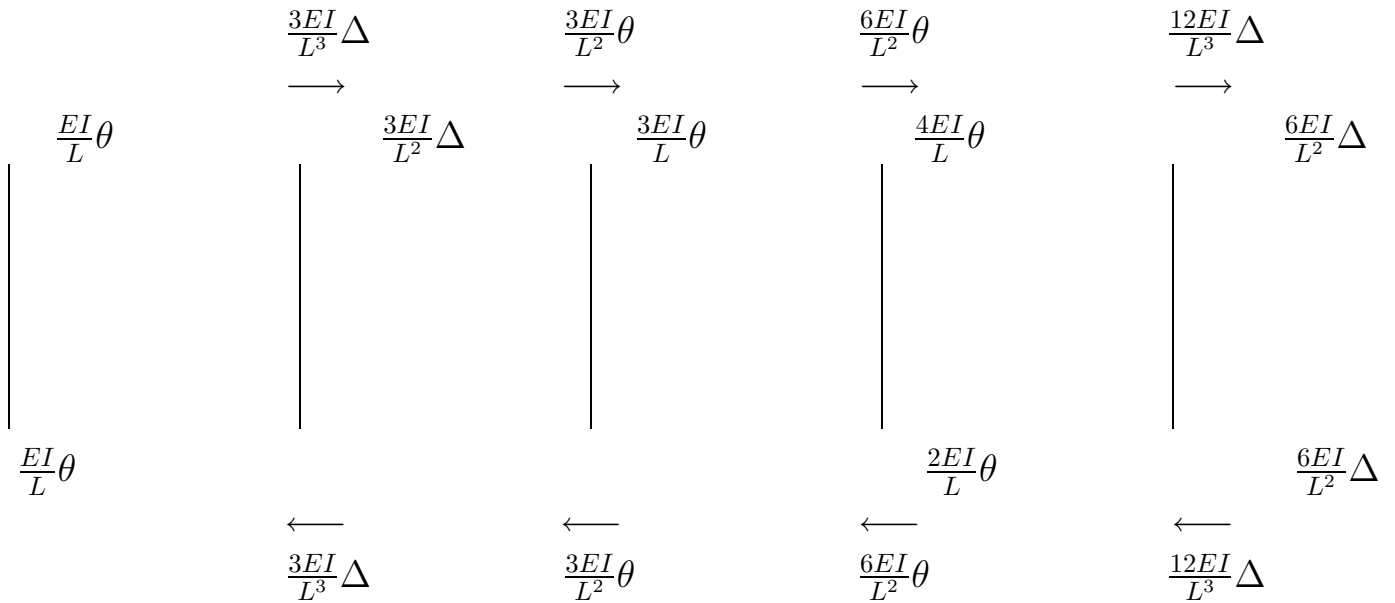
$$\Phi = \frac{12EI}{G(A/\alpha)L^2} = 24\alpha(1 + \nu) \left(\frac{r}{L}\right)^2$$

and r is the “radius of gyration” of the cross section, $r = \sqrt{I/A}$.

To neglect shear deformation, set $\Phi = 0$:

$$\mathbf{k} = \begin{bmatrix} \frac{EA}{L} & 0 & 0 \\ 0 & \frac{4EI}{L} & \frac{2EI}{L} \\ 0 & \frac{2EI}{L} & \frac{4EI}{L} \end{bmatrix}$$

Beam Element Stiffness Matrix in Local Coordinates: k



$$\begin{Bmatrix} N_1 \\ V_1 \\ M_1 \\ N_2 \\ V_2 \\ M_2 \end{Bmatrix} = \begin{bmatrix} \frac{EA}{L} & 0 & 0 & -\frac{EA}{L} & 0 & 0 \\ & \frac{12EI}{L^3} & \frac{6EI}{L^2} & 0 & -\frac{12EI}{L^3} & \frac{6EI}{L^2} \\ & & \frac{4EI}{L} & 0 & -\frac{6EI}{L^2} & \frac{2EI}{L} \\ & & & \frac{EA}{L} & 0 & 0 \\ & & & & \frac{12EI}{L^3} & -\frac{6EI}{L^2} \\ & & & & & \frac{4EI}{L} \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \\ u_5 \\ u_6 \end{Bmatrix}$$

SYM

Relationships between Local Coordinates and Global Coordinates: \mathbf{T}

The geometric relationship between local displacements, \mathbf{u} , and global displacements, \mathbf{v} , is

$$u_1 = v_1 \cos \theta + v_2 \sin \theta \quad u_2 = -v_1 \sin \theta + v_2 \cos \theta \quad u_3 = v_3$$

or, $\mathbf{u} = \mathbf{T} \mathbf{v}$.

The equilibrium relationship between local forces, \mathbf{q} , and global forces, \mathbf{f} , is

$$q_1 = f_1 \cos \theta + f_2 \sin \theta \quad q_2 = -f_1 \sin \theta + f_2 \cos \theta \quad q_3 = f_3$$

or, $\mathbf{q} = \mathbf{T} \mathbf{f}$, where, in both cases,

$$\mathbf{T} = \begin{bmatrix} c & s & 0 & & & \\ -s & c & 0 & & 0 & \\ 0 & 0 & 1 & & & \\ & & & c & s & 0 \\ 0 & -s & c & 0 & & \\ & & & 0 & 0 & 1 \end{bmatrix} \quad c = \cos \theta = \frac{x_2 - x_1}{L}$$

$$s = \sin \theta = \frac{y_2 - y_1}{L}$$

The coordinate transformation matrix, \mathbf{T} , is orthogonal, $\mathbf{T}^{-1} = \mathbf{T}^T$.

Fixed-Pinned Beam, k

$$\begin{bmatrix} \frac{EA}{L} & 0 & 0 & -\frac{EA}{L} & 0 & 0 \\ & \frac{3EI}{L^3} & \frac{3EI}{L^2} & 0 & -\frac{3EI}{L^3} & 0 \\ & & \frac{3EI}{L} & 0 & -\frac{3EI}{L^2} & 0 \\ & & & \frac{EA}{L} & 0 & 0 \\ \text{SYM} & & & & \frac{3EI}{L^3} & 0 \\ & & & & & 0 \end{bmatrix}$$

Pinned-Fixed Beam, k

$$\begin{bmatrix} \frac{EA}{L} & 0 & 0 & -\frac{EA}{L} & 0 & 0 \\ & \frac{3EI}{L^3} & 0 & 0 & -\frac{3EI}{L^3} & \frac{3EI}{L^2} \\ & & 0 & 0 & 0 & 0 \\ & & & \frac{EA}{L} & 0 & 0 \\ \text{SYM} & & & & \frac{3EI}{L^3} & -\frac{3EI}{L^2} \\ & & & & & \frac{3EI}{L} \end{bmatrix}$$

For beam element stiffness matrices including shear deformations, see:
Theory of Matrix Structural Analysis, by J.S. Przemieniecki (Dover Pub., 1985).
 (... a steal at \$12.95)

Fixed-Pinned Beam in Global Coordinates, \mathbf{K}

$$\mathbf{K} = \begin{bmatrix} \frac{EA}{L}c^2 & \frac{EA}{L}cs & -\frac{3EI}{L^2}s & -\frac{EA}{L}c^2 & -\frac{EA}{L}cs & 0 \\ +\frac{3EI}{L^3}s^2 & -\frac{3EI}{L^3}cs & & -\frac{3EI}{L^3}s^2 & +\frac{3EI}{L^3}cs & \\ & \frac{EA}{L}s^2 & \frac{3EI}{L^2}c & -\frac{EA}{L}cs & -\frac{EA}{L}s^2 & 0 \\ & +\frac{3EI}{L^3}c^2 & & +\frac{3EI}{L^3}cs & -\frac{3EI}{L^3}c^2 & \\ & & \frac{3EI}{L} & \frac{3EI}{L^2}s & -\frac{3EI}{L^2}c & 0 \\ & & & \frac{EA}{L}c^2 & \frac{EA}{L}cs & 0 \\ & & & +\frac{3EI}{L^3}s^2 & -\frac{3EI}{L^3}cs & \\ & & \text{SYM} & & & \\ & & & & \frac{EA}{L}s^2 & 0 \\ & & & & +\frac{3EI}{L^3}c^2 & \\ & & & & & 0 \end{bmatrix}$$

Pinned-Fixed Beam in Global Coordinates, \mathbf{K}

$$\mathbf{K} = \begin{bmatrix} \frac{EA}{L}c^2 & \frac{EA}{L}cs & 0 & -\frac{EA}{L}c^2 & -\frac{EA}{L}cs & -\frac{3EI}{L^2}s \\ +\frac{3EI}{L^3}s^2 & -\frac{3EI}{L^3}cs & & -\frac{3EI}{L^3}s^2 & +\frac{3EI}{L^3}cs & \\ & \frac{EA}{L}s^2 & 0 & -\frac{EA}{L}cs & -\frac{EA}{L}s^2 & \frac{3EI}{L^2}c \\ & +\frac{3EI}{L^3}c^2 & & +\frac{3EI}{L^3}cs & -\frac{3EI}{L^3}c^2 & \\ & & 0 & 0 & 0 & 0 \\ & & & \frac{EA}{L}c^2 & \frac{EA}{L}cs & \frac{EI}{L^2}s \\ & & & +\frac{3EI}{L^3}s^2 & -\frac{3EI}{L^3}cs & \\ & & \text{SYM} & & & \\ & & & & \frac{EA}{L}s^2 & -\frac{3EI}{L^2}c \\ & & & & \frac{3EI}{L^3}c^2 & \\ & & & & & \frac{3EI}{L} \end{bmatrix}$$

Notation

- u** = Element deflection vector in the Local coordinate system
q = Element force vector in the Local coordinate system
k = Element stiffness matrix in the Local coordinate system
 ... $\mathbf{q} = \mathbf{k} \mathbf{u}$
- T** = Coordinate Transformation Matrix
 ... $\mathbf{T}^{-1} = \mathbf{T}^T$
- v** = Element deflection vector in the Global coordinate system
 ... $\mathbf{u} = \mathbf{T} \mathbf{v}$
- f** = Element force vector in the Global coordinate system
 ... $\mathbf{q} = \mathbf{T} \mathbf{f}$
- K** = Element stiffness matrix in the Global coordinate system
 ... $\mathbf{K} = \mathbf{T}^T \mathbf{k} \mathbf{T}$
- d** = Structural deflection vector in the Global coordinate system
p = Structural load vector in the Global coordinate system
K_s = Structural stiffness matrix in the Global coordinate system
 ... $\mathbf{p} = \mathbf{K}_s \mathbf{d}$

	Local	Global
Element Deflection	u	v
Element Force	q	f
Element Stiffness	k	K
Structural Deflection	-	d
Structural Loads	-	p
Structural Stiffness	-	K_s